Astrophysically relevant geodesics in Kerr spacetime

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Outline

1 Kerr Spacetime

- 2 3D timelike geodesics
- 3 Analytic solutions
- 4 Application and summary

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Space-time around a rotating black hole

Kerr metric: (*Kerr R. 1963, PRL, 11, 237–238;* Kerr 1963)

$$ds^{2} = -(1 - \frac{2M_{\bullet}r}{\rho^{2}})dt^{2} + \frac{\rho^{2}}{\Delta}dr^{2} + (r^{2} + a^{2} + \frac{2M_{\bullet}ra^{2}\sin^{2}\theta}{\rho^{2}})\sin^{2}\theta d\phi^{2} - \frac{4M_{\bullet}ar\sin^{2}\theta}{\rho^{2}} d\phi dt + \rho^{2}d\theta^{2}$$
(1)

{*t*, *r*, θ , ϕ } - Boyer Lindquist coordinates, G = c = 1, geometrical units, $\rho^2 = r^2 + a^2 \cos^2 \theta$, $\Delta \equiv r^2 - 2M_{\bullet}r + a^2$, $a \equiv J/M_{\bullet}$ - spin.

Conserved quantities for timelike geodesics:

- Energy per unit rest mass: *E*, stationary metric.
- z-component of the angular momentum: *L*, axis symmetric metric.
- 3 <u>Carter's constant</u>: Q (Carter 1968, Phys. Rev. D, 174 1559-71; Carter 1968) Separability r and θ components of the Hamilton-Jacobi equation.

History of timelike Kerr geodesics

- <u>Carter 1968, PRD</u>: Pioneer work on solving Hamilton-Jacobi equation of motion to get quadrature form.
- Wilkins 1972, PRD: Solution for spherical orbits for extreme spin, a = 1.
- Bardeen et al 1972, ApJ: Deriving {*E*, *L*} for circular orbits and ISCO radius.
- Schmidt 2002, CQG: Deriving fundamental orbital frequencies $\{\nu_{\phi}, \nu_{r}, \nu_{\theta}\}$ in quadrature form.
- Fujita & Hikida 2009, CQG: First fully-analytic form of equations of motion.
- Rana & Mangalam 2019, CQG: Simpler analytic form of equations of motion using eccentricity and inverse-latum rectum of orbit {e, μ}, and study of various types of orbits.
- G. Compère et al 2022, PRD: Detailed study of classification of radial motion.

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3D Kerr trajectories

- Axis symmetry of the spacetime leads to change in plane of the orbit.
- Non-equatorial eccentric bound orbits, spherical orbits, and non-equatorial separatrix orbits. (Rana P. & A. Mangalam 2019, Class. Quant. Grav. 36, 045009, arXiv:1901.02730, RM19).



Figure: 1. (a) A non-equatorial eccentric trajectory, (b) a spherical trajectory, and (c) a separatrix trajectory; RM19.

Astrophysically relevant geodesics in Kerr spacetime 3D timelike geodesics

Motivation to study Kerr orbit dynamics

 <u>QPOs in BHXRB</u>: Study of Quasi-periodic oscillations observed in Fourier power density spectrum of Black hole X-ray binaries, indicating fundamental frequencies of oscillations near a black hole.



Figure: 2. (a) Black hole X-ray binary, *Courtesy: Web*; (b) Simultaneous QPOs in the PSD of GRO J1655-40 using RXTE data. *Courtesy: T.E.Strohmayer 2001, ApJ, 552, L49;* (c) A cartoon of an EMRI *Courtesy: Web.*

 <u>GWs from EMRIs</u>: Extreme-mass ratio inspirals follow a series of Kerr orbits before plunging onto the supermassive black hole.

Radial motion

The effective potential for the radial motion



Roots of the radial equation using Ferrari's method (*G. Cardano, The rules of algebra. Translated by T. R. Witmer 1968; Cardano 1968*)



Figure: 3. Radial effective potentials for various radial motions.

Trajectory solutions (RM19)

Equations for time-like geodesics (Carter 1968, Schmidt 2002)

$$\phi - \phi_0 = -\frac{1}{2} \int_{r_0}^r \frac{1}{\Delta\sqrt{R}} \frac{\partial R}{\partial L} dr' - \frac{1}{2} \int_{\theta_0}^{\theta} \frac{1}{\sqrt{\Theta}} \frac{\partial \Theta}{\partial L} d\theta' = -\frac{1}{2} I_1 - \frac{1}{2} H_1, \quad (3a)$$

$$t - t_0 = \frac{1}{2} \int_{r_0}^r \frac{1}{\Delta\sqrt{R}} \frac{\partial R}{\partial E} dr' + \frac{1}{2} \int_{\theta_0}^{\theta} \frac{1}{\sqrt{\Theta}} \frac{\partial \Theta}{\partial E} d\theta' = \frac{1}{2} I_2 + \frac{1}{2} H_2, \quad (3b)$$

$$\int_{r_0}^r dr' = \int_{r_0}^{\theta} \frac{\partial \theta'}{\partial E} d\theta' + I_0 = H_0 \quad (3c)$$

$$\int_{0}^{s} \frac{dr}{\sqrt{R}} = \int_{\theta_0}^{s} \frac{d\theta}{\sqrt{\Theta}} \Rightarrow I_8 = H_3,$$
(3c)

where $R(r) \ge 0$ and $\Theta(\theta) \ge 0$.

],

Substituting $r_1 = 1/[\mu(1-e)]$ and $r_2 = 1/[\mu(1+e)]$ removes the singularity of integrands at turning points:

$$R = (E^{2} - 1) (r - r_{1}) (r - r_{2}) (r - r_{3}) (r - r_{4}).$$
(4)

(e, μ, a, Q) parameter space gives simpler analytic compact forms of the integrals I₁, I₂, I₈, H₁, H₂, and H₃.

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Analytic forms for trajectory solutions (RM19)

$$\phi - \phi_0 = -\frac{1}{2}I_1 - \frac{1}{2}H_1 = \frac{1}{2}\left[C_3I_3\left(\alpha, e, \mu, a, Q\right) + C_4I_4\left(\alpha, e, \mu, a, Q\right) - H_1\left(\theta, \theta_0, e, \mu, a, Q\right)\right],$$
(5a)

$$t - t_0 = \frac{1}{2}I_2 + \frac{1}{2}H_2 = \frac{1}{2}\left[C_5I_5(\alpha, e, \mu, a, Q) + C_6I_6(\alpha, e, \mu, a, Q) + C_7I_3(\alpha, e, \mu, a, Q)\right]$$
(5b)

$$+C_{8}I_{4}(\alpha, e, \mu, a, Q) + H_{2}(\theta, \theta_{0}, e, \mu, a, Q)],$$
(5b)

 $I_8(\alpha, e, \mu, a, Q) = H_3(\theta, \theta_0, e, \mu, a, Q),$

(5c)

Analytic forms for trajectory solutions (RM19)

Integrals	Analytic forms of the θ integrals	
H_1	$\left \frac{2L}{z_{+}a\sqrt{1-E^{2}}}\left\{\Pi\left(z_{-}^{2}, \arcsin\left(\frac{\cos\theta}{z_{-}}\right), \frac{z_{-}^{2}}{z_{+}^{2}}\right) - \Pi\left(z_{-}^{2}, \arcsin\left(\frac{\cos\theta_{0}}{z_{-}}\right), \frac{z_{-}^{2}}{z_{+}^{2}}\right) - F\left(\arcsin\left(\frac{\cos\theta}{z_{-}}\right), \frac{z_{-}^{2}}{z_{+}^{2}}\right) + F\left(\arcsin\left(\frac{\cos\theta_{0}}{z_{-}}\right), \frac{z_{-}^{2}}{z_{+}^{2}}\right)\right\}$	
H_2	$\frac{2Eaz_{+}}{\sqrt{1-E^{2}}}\left\{K\left(\arcsin\left(\frac{\cos\theta}{z_{-}}\right),\frac{z_{-}^{2}}{z_{+}^{2}}\right)-K\left(\arcsin\left(\frac{\cos\theta_{0}}{z_{-}}\right),\frac{z_{-}^{2}}{z_{+}^{2}}\right)-F\left(\arcsin\left(\frac{\cos\theta}{z_{-}}\right),\frac{z_{-}^{2}}{z_{+}^{2}}\right)+F\left(\arcsin\left(\frac{\cos\theta_{0}}{z_{-}}\right),\frac{z_{-}^{2}}{z_{+}^{2}}\right)\right\}$	
H ₃	$\frac{1}{a\sqrt{1-E^2}z_+}\left\{F\left(\arcsin\left(\frac{\cos\theta_0}{z}\right),\frac{z^2}{z_+^2}\right)-F\left(\arcsin\left(\frac{\cos\theta}{z}\right),\frac{z^2}{z_+^2}\right)\right\}$	

Table: 1. Analytic expressions for integrals involved in the equations of motion.

where m^2 , p_1^2 , p_2^2 , p_3^2 , k^2 , s^2 , *A*, *B*, *C*, z_- , z_+ , and C_3 - C_8 are functions of parameters (*e*, μ , *a*, *Q*).

- Advantage of novel trajectory solution: Removes the singularity at the turning points of orbits and provides faster computation.
- Standard Elliptic integrals: $F(\alpha, k^2), K(\alpha, k^2), \Pi(s^2, \alpha, k^2)$

Bound orbit conditions in (*e*, μ , *a*, *Q*) space (RM19)

Condition for the real integrals:

$$\begin{bmatrix} \mu^3 a^2 Q (1+e)^2 + \mu^2 (\mu a^2 Q - x^2 - Q) \\ \times (3-e) (1+e) + 1 \end{bmatrix} \ge 0. \quad (6a)$$

Periastron point is outside horizon:

$$\left[\mu\left(1+e\right)\left(1+\sqrt{1-a^2}\right)\right] < 1. \tag{6b}$$

Bound orbit between outermost roots of the effective potential:

$$E(e, \mu, a, Q) < 1.$$
 (6c)



Figure 4. The bound orbit region in the

(e, μ) plane for a = 0.5 and Q = 5.

The red curve represents separatrix orbits. S represents ISSO, and M represents MBSO.

ISSO and MBSO



Figure: 5. The horizon radius (r_+ , red), light radius [green ring], MBSO (blue), and ISSO (black) (embedded in flat space), (a) side view, (b) top view, are shown, where {a = 0.5, Q = 4}.

Separatrix trajectories (RM19)



Figure: 7. A Separatrix trajectory with $\{e = 0.6, r_p = 4.075R_g, a = 0.2, Q = 3\}$ satisfying Eq. (8).

 Every inspiraling orbit passes through a separatrix orbit on the transition from bound to plunge. Hence, they are important in the study of inspiralling bodies, for example, study of GW emission from extreme-mass ratio inspirals (EMRIs).

Separatrix trajectories (RM19)

Position of separatrix in the parameter space:

е

$$s_{s}(r_{s}, a, Q) = \frac{4 + a' r_{s} + \sqrt{(r_{s}a' + 2)^{2} - 4d' r_{s}^{4}}}{-a' r_{s} - \sqrt{(r_{s}a' + 2)^{2} - 4d' r_{s}^{4}}},$$
(7a)

$$\mu_{s}(r_{s}, a, Q) = \frac{1}{4r_{s}} \left[-a'r_{s} - \sqrt{\left(r_{s}a' + 2\right)^{2} - 4d'r_{s}^{4}} \right];$$
(7b)

where a' and d' are functions of (r_s, a, Q) . These formulae reduce to the equatorial case, Q = 0, previously derived by *Levin & Perez-Giz*, 2009, *PRD*, 79(12), 124013.

Trajectory solutions: Using equation

$$\mu^{3}a^{2}Q(1+e)^{2} + \mu^{2}\left(\mu a^{2}Q - x^{2} - Q\right)(3-e)(1+e) + 1 = 0,$$
(8)

we obtained trajectory solutions in terms of the logarithmic and trigonometric functions.

Astrophysically relevant geodesics in Kerr spacetime Analytic solutions

Equatorial eccentric orbits ($e \neq 0$, Q = 0) and spherical orbits (e = 0, $Q \neq 0$) (RM19)



Figure: 6. (a) An equatorial eccentric trajectory, and (b) a spherical orbit near a rotating black hole.

Energy and Angular momentum of spherical orbits (RM19)

General expression for *E* and *L* for spherical orbits as functions of (*r*_s, *a*, *Q*):

$$E(r_{\rm s}, a, Q) = \frac{\begin{cases} 2a^4Q + (r_{\rm s} - 3)(r_{\rm s} - 2)^2 r_{\rm s}^4 - a^2r_{\rm s}\left[r_{\rm s}^2(3r_{\rm s} - 5) + Q(r_{\rm s}(r_{\rm s} - 4) + 5)\right] \\ -2a\left[r_{\rm s}(r_{\rm s} - 2) + a^2\right]\sqrt{a^2Q^2 - r_{\rm s}^3Q(r_{\rm s} - 3) + r_{\rm s}^5} \end{cases}^{1/2}}{r_{\rm s}^2\left[r_{\rm s}(r_{\rm s} - 3)^2 - 4a^2\right]^{1/2}}, \quad (9a)$$

$$x(r_{s}, a, Q) = \frac{\begin{cases} -2a^{4}Q + r_{s}^{2} (r_{s} - 3) \left[r_{s}^{2} - (r_{s} - 3) Q \right] + a^{2}r_{s} \left(r_{s}^{3} + r_{s}^{2} - 2Qr_{s} + 8Q \right) \\ \frac{-2a \left[r_{s} (r_{s} - 2) + a^{2} \right] \sqrt{a^{2}Q^{2} - r_{s}^{3}Q(r_{s} - 3) + r_{s}^{5}} \end{cases}}{r_{s}^{1/2} \left[r_{s} (r_{s} - 3)^{2} - 4a^{2} \right]^{1/2}}, \quad (9b)$$

and

$$L(r_s, a, Q) = x(r_s, a, Q) + aE(r_s, a, Q).$$
(9c)

These expressions reduce to the standard formulae of (*E*, *L*) for equatorial circular orbits, *Q* = 0 (*Bardeen et. al. 1972, ApJ., 178, 347-370*).

Fundamental frequencies (RM19)

- We also derive the fundamental frequencies $(\nu_{\phi}, \nu_{r}, \nu_{\theta})$ using a long time average method.
- Exact expressions of ν_{ϕ} , ν_r , and ν_{θ} for non-equatorial eccentric trajectories:

$$\nu_{\phi} = \frac{c^{3} \left\{ \left[-l_{1}\left(\frac{\pi}{2}, e, \mu, a, Q\right) - 2Ll_{8}\left(\frac{\pi}{2}, e, \mu, a, Q\right) \right] F\left(\frac{\pi}{2}, \frac{z^{2}_{-}}{z^{2}_{+}}\right) + 2Ll_{8}\left(\frac{\pi}{2}, e, \mu, a, Q\right) \Pi\left(z^{2}_{-}, \frac{\pi}{2}, \frac{z^{2}_{-}}{z^{2}_{+}}\right) \right\}}{2\pi GM_{\bullet}f(e, \mu, a, Q)}, \qquad (10a)$$

$$\nu_{r} = \frac{c^{3}F\left(\frac{\pi}{2}, \frac{z^{2}_{-}}{z^{2}_{+}}\right)}{GM_{\bullet}f(e, \mu, a, Q)}, \qquad \nu_{\theta} = \frac{c^{3}a\sqrt{1 - E^{2}}z_{+}I_{8}\left(\frac{\pi}{2}, e, \mu, a, Q\right)}{2GM_{\bullet}f(e, \mu, a, Q)}, \qquad (10b)$$

To summarize:

Our analytic results have made advancements to various orbital dynamics study done by Bardeen et. al. 1972, Schmidt 2002, Levin & Perez-Giz 2009, and Fujita & Hikida 2009 in the Kerr space-time.

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Summary

	Results presented in the previous studies	New results
1.	<u>Bardeen1972</u> : Analytic expressions of <i>E</i> and <i>L</i> for equatorial circular orbits.	Spherical orbits : Analytic expressions for <i>E</i> and <i>L</i> and important radii - ISSO, MBSO and general spherical orbits with $Q \neq 0$.
2.	<u>Schmidt2002</u> : Expressions for fundamental frequencies, $(\nu_{\phi}, \nu_{\tau}, \nu_{\theta})$, of general trajectories given in terms of quadratures using action-angle variables.	Frequencies: t , ϕ , $r - \theta$ integrals are solved analytically and explicit expressions for $(\nu_{\phi}, \nu_{r}, \nu_{\theta})$ are presented for $Q \neq 0$ using long-time average method.
3.	<i>Fujita2009</i> : Analytic trajectory solutions are presented as a function of the variable Mino time, λ , for $Q \neq 0$, where input variables are <i>E</i> , L_z , <i>Q</i> and <i>a</i> .	Trajectory : Analytic solutions are presented which provide faster computation using (e, μ , a, Q) space. - Bound orbit conditions in {e, μ , a, Q} and {E, L, a, Q} spaces are derived. - Simpler trajectory solutions are obtained for equatorial eccentric and spherical orbits. - Translation relations are obtained between {E, L, a, Q} and {e, μ , a, Q} parameter spaces.
4.	<u>Levin2009</u> : Expressions for the eccentricity, e_s , and inverse latus-rectum, μ_{s} , of equatorial separatrix orbits and the trajectory solutions ($Q = 0$) are presented.	Separatrix: Expressions for e_s and μ_s and trajectory solutions are presented for general separatrix orbits with $Q \neq 0$.

Table: 2. A list of novel formulae and advancement in the theoretical results.

Applications

- Rana, P. & Mangalam, A., 2020a, "A Geometric Origin for Quasi-periodic Oscillations in Black Hole X-Ray Binaries", ApJ., 903(2), 121, arXiv:2009.01832; RM20a.
- Rana, P. & Mangalam, A., 2020b, "A Relativistic Orbit Model for Temporal Properties of AGN", Galaxies, 8(3), 67, arXiv:2009.03061; RM20b.